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OMB No. 0704-0188

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5c. PROGRAM ELEMENT NUMBER	
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5d. PROJECT NUMBER	
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5e. TASK NUMBER	
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## 8. PERFORMING ORGANIZATION REPORT

10. SPONSOR/MONITOR'S ACRONYM(S)

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NUMBER(S)

NUMBER(S) Please see attached

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OF PAGES

19a. NAME OF RESPONSIBLE PERSON

Leilani Richardson

a. REPORT

b. ABSTRACT

**c. THIS PAGE**

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19b. TELEPHONE NUMBER

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MEMORANDUM FOR PR (In-House Publication)

FROM: PROI (TI) (STINFO)

05 Jun 2000

SUBJECT: Authorization for Release of Technical Information, Control Number: **AFRL-PR-ED-TP-2000-123**  
Harper, J., "Progress in Advanced Propellant Research"

**NASA JPL-MSFC 11<sup>th</sup> Advances Space Propulsion Research Workshop** (Statement A)  
**(Pasadena, CA, 31 May-02 Jun 00)** (Submission Deadline: 02 Jun 00)

1. This request has been reviewed by the Foreign Disclosure Office for: a.) appropriateness of distribution statement, b.) military/national critical technology, c.) export controls or distribution restrictions, d.) appropriateness for release to a foreign nation, and e.) technical sensitivity and/or economic sensitivity.

Comments: \_\_\_\_\_  
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Signature \_\_\_\_\_ Date \_\_\_\_\_

2. This request has been reviewed by the Public Affairs Office for: a.) appropriateness for public release and/or b) possible higher headquarters review.

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3. This request has been reviewed by the STINFO for: a.) changes if approved as amended, b.) appropriateness of distribution statement, c.) military/national critical technology, d.) economic sensitivity, e.) parallel review completed if required, and f.) format and completion of meeting clearance form if required

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4. This request has been reviewed by PR for: a.) technical accuracy, b.) appropriateness for audience, c.) appropriateness of distribution statement, d.) technical sensitivity and economic sensitivity, e.) military/national critical technology, and f.) data rights and patentability

Comments: \_\_\_\_\_  
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APPROVED/APPROVED AS AMENDED/DISAPPROVED

\_\_\_\_\_  
ROBERT C. CORLEY (Date)  
Senior Scientist (Propulsion)  
Propulsion Directorate

# Progress in Advanced Propellant Research

- Cryogenic HEDM Solids
- Polynitrogen Compounds
- High Performance Monopropellants
- Hydrocarbon Fuels

Capt. Jessica Harper

Air Force Research Laboratory, Propellants Branch

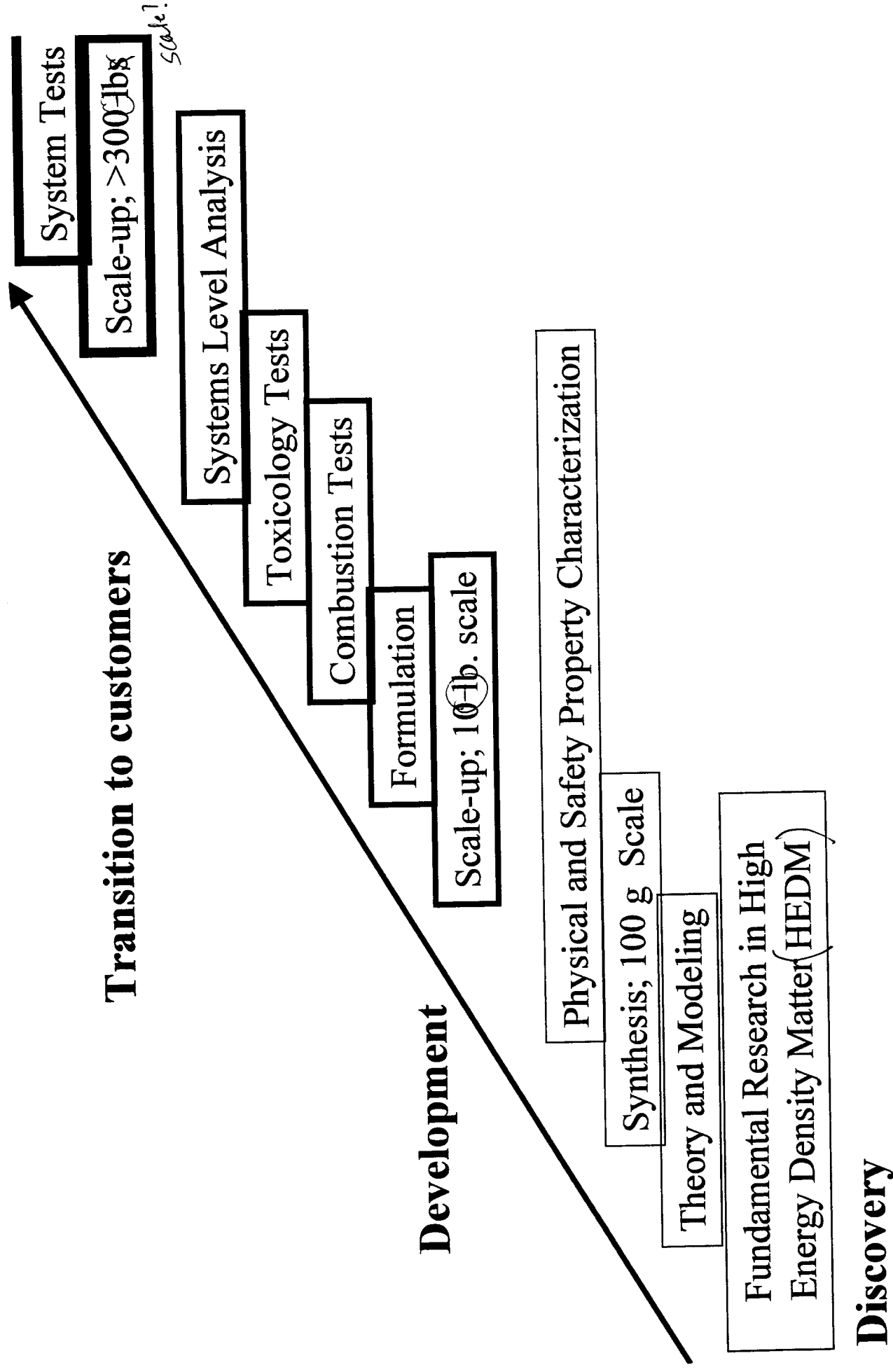
Edwards AFB, CA 93524-7680

(661) 275-5883; [jessica.harper@ple.af.mil](mailto:jessica.harper@ple.af.mil)

Propulsion Directorate mission includes discovery and development of evolutionary and revolutionary advances in propulsion systems

*Distribution A: Unlimited Release  
(I have added this to my cover slide. gH)*

# Technology Development Path



# Cryogenic Solid HEDM Propellants

Use a solidified fuel or oxidizer as a storage medium for energetic additives, obtaining density and specific-impulse improvements

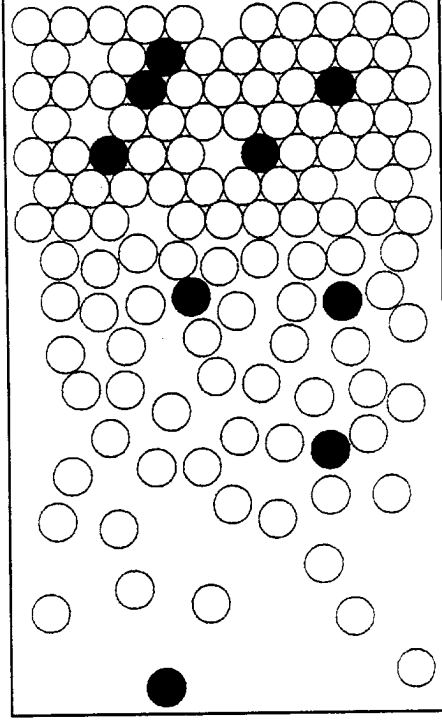
Depositing certain atomic or molecular species in solid hydrogen at 5% concentrations can increase specific impulse by more than 20%

Vehicle	Propellant	Payload Mass (lb)	Payload (lb) With 10% Density Increase	Payload (lb) With 10% Isp Increase	Payload (lb) With 10% Increase in Both
Rockwell SSTO RLV	LH2/LOX (Isp = 455 s)	40,000	51,200 (+28%)	68,000 (+70%)	76,800 (+92%)

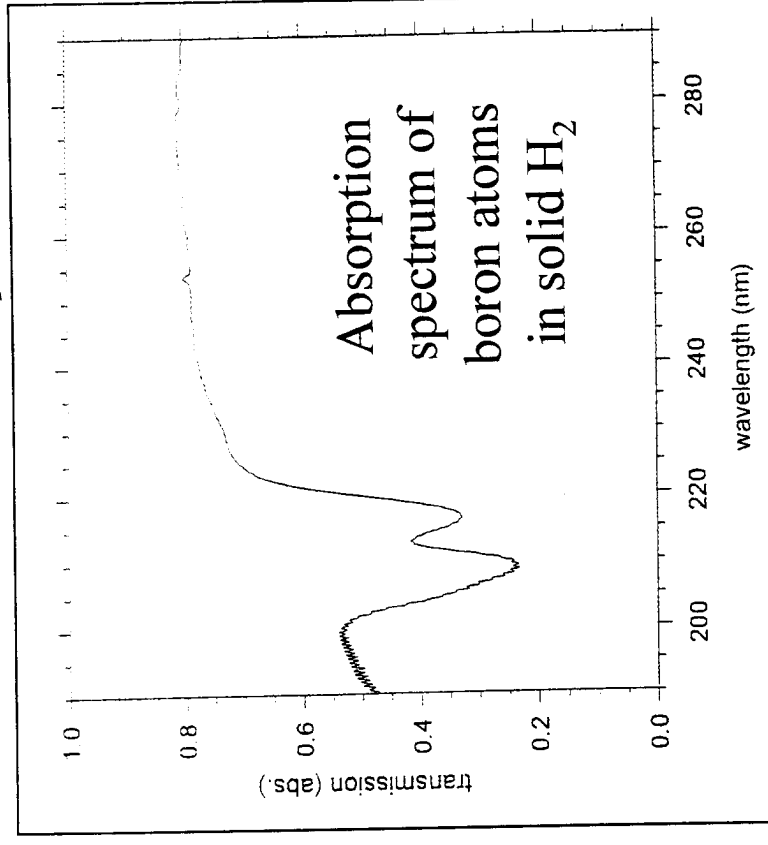
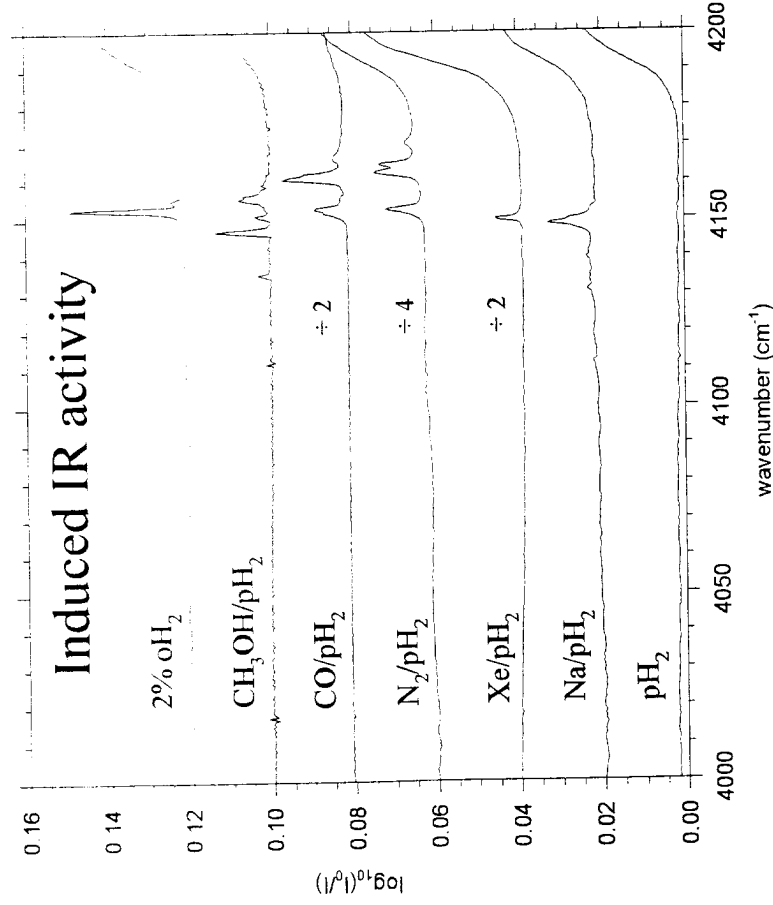
Large payload increases are achievable with modest density or specific impulse increases

# Cryogenic Solid HEDM Propellants

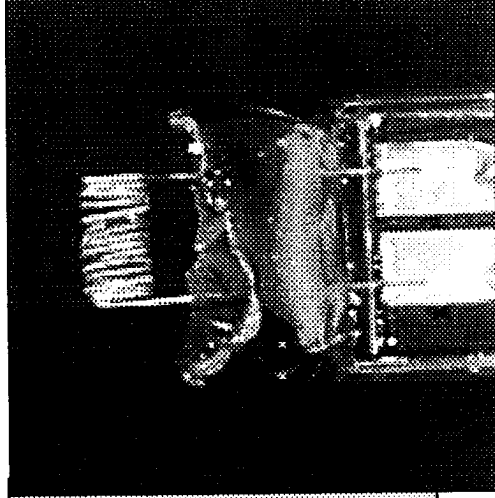
- **KEY ISSUE** Develop methods to characterize thick cryogenic solids with high additive concentrations
- **APPROACH** Investigate how direct dopant absorptions and the dopant-induced IR activity of  $H_2$  relate to the type and concentration of HEDM additives.



Schematic of the deposition process



# Cryogenic Solid HEDM Prospectus



- *KEY ISSUE* Develop and characterize high-flux, robust, pure sources of desirable HEDM additives
- *APPROACH*  
Boron filament source  
Commercial aluminum deposition source

- *KEY ISSUE* Determine reactivity, diffusion, and recombination propensities to identify the most stable additives *toxic?*
- *APPROACH* Simulations and experiments are yielding significant information about the chemical and thermal stabilities of doped solid  $H_2$  and the dopant concentration limits.

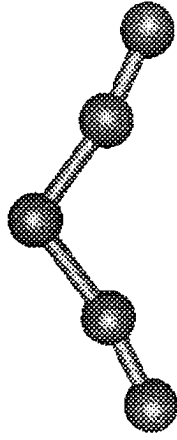
*Substrate?*

BASELINE	Isp (sec)
Solid $H_2$	389
5% ADDITIVE	
H	407
Be	452
B	472
C	469
Al	425
BH	449
CH	433
LiB	494
LiC	465
B <sub>2</sub>	492
BC <sub>2</sub>	482
B <sub>2</sub> C <sub>2</sub>	439

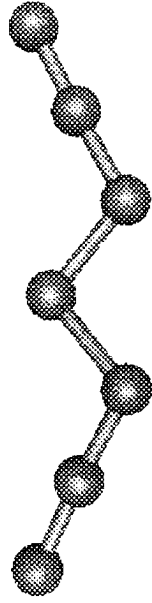
*(10) 100% H<sub>2</sub> HEDM*

# Polynitrogen Calculations Led to Synthesis

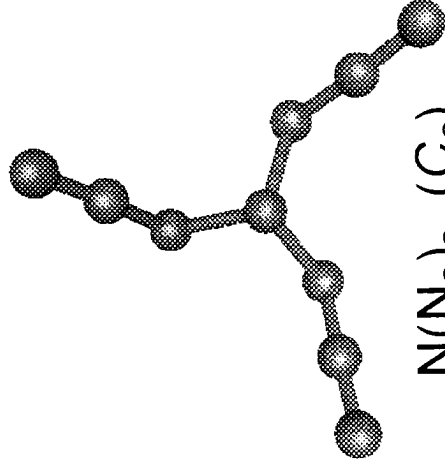
**Program Objective: Synthesize and characterize new highly energetic polynitrogen compounds**



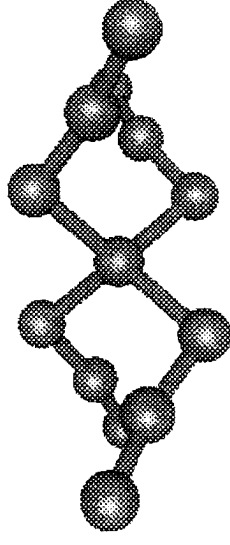
$\text{N}_5^+ (C_{2v})$



$\text{N}(\text{N}_3)_2^- (C_2)$



$\text{N}(\text{N}_3)_3 (C_3)$



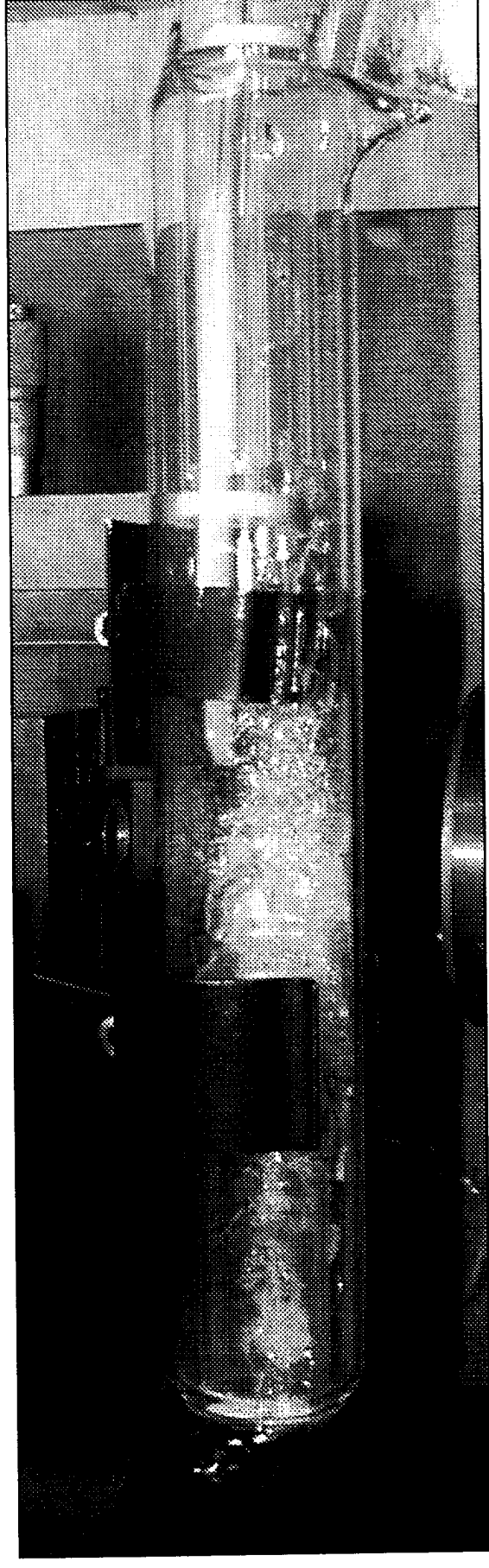
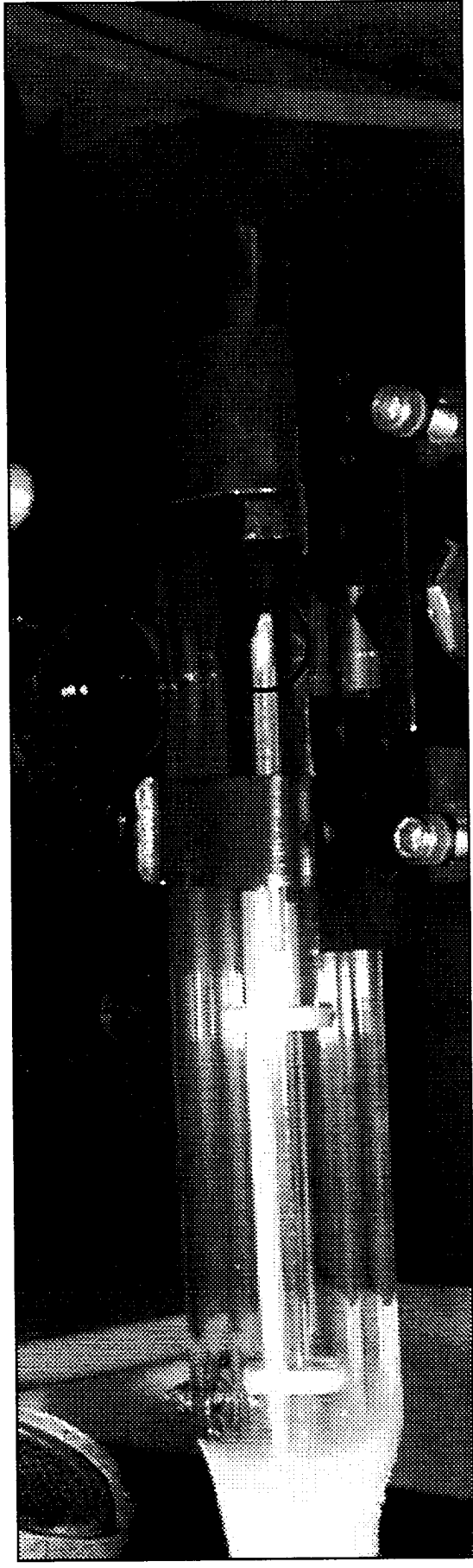
$\text{N}(\text{N}_3)_4^+ (D_{2d})$

- Calculations showed that the compounds depicted above are all stable, albeit to varying extents, and highly energetic
- Predicted measurable properties to help chemists identify what they made

This project is co-funded by DARPA and AFOSR



# $\text{N}_5^+$ Salt in Low-Temperature Raman Spectrometer: Before and After Explosion



# All Nitrogen Compound

Program Goal: Discover, characterize, and scale-up production of highly energetic all-nitrogen compounds for use as propellants or explosives.

Drs. Karl Christe, William Wilson, Jeffrey Sheehy, Jerry Boatz

## Recent Accomplishments

- The  $N_5^+$  cation is the first all-nitrogen compound to be prepared in bulk in over 100 years, and it is only the third known compound of this type

- Improved synthesis yields 5 g of  $N_5^+$  salt in high purity at a fraction of the time of the previous synthesis

- A remarkably stable  $N_5^+SbF_6^-$  salt was unambiguously identified by vibrational and NMR spectroscopy of unlabeled and  $^{15}N$ -labeled  $N_5^+$  in conjunction with quantum-chemical calculations

- Based on this initial success, the synthesis of additional new polynitrogen compounds is planned. A particular target is  $N_8$ , which could be formed as the ionic salt  $N_5^+N_3^-$ , or as a covalent compound such as azidopentazole

## Significance

- Named one of "Chemistry's Top Five Achievements in 1999" by the American Chemical Society
- The feasibility of polynitrogen-based HEDM compounds has been demonstrated by the successful bulk synthesis of  $N_5^+$
- Demonstrates synergy of calculations and experiment. Modeling and simulation were used to predict that the compound should exist, and computed spectra made it possible to identify and characterize the new  $N_5^+$  compound
- Stable, neutral polynitrogens would be high performance monopropellants or explosives.  $N_8$  as a monopropellant would have an Isp of 420 s, compared with hydrazine at 233 s. Clean propellant would have cost savings in handling, production, logistics, and disposal.

# Monopropellant Development

## Objective:

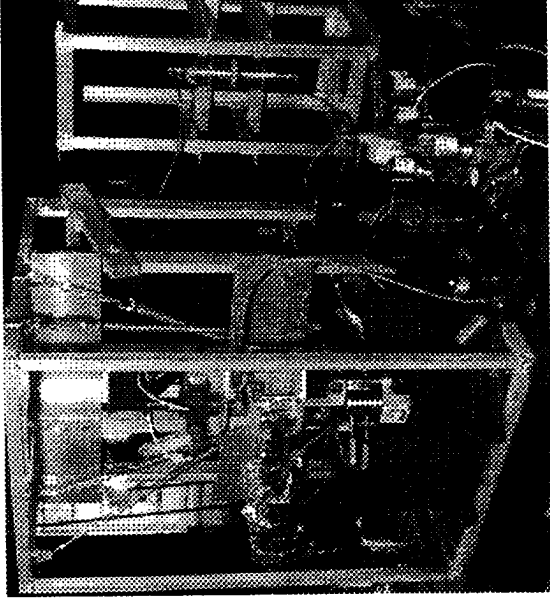
- 1) Develop monopropellants with higher performance and lower toxicity than SOTA (hydrazine).
- 2) Develop monopropellant to exceed bipropellant performance

## Approach:

- 1) Synthesize and characterize energetic liquid salt ingredients
- 2) Formulate monopropellants with new ingredient
- 3) Combustion testing of formulations

## Motives:

- 1) Enable AF satellite missions; AF aerospace vehicle; reduced size boost vehicle
- 2) Enable NASA small (< 100 kg) spacecraft and SSTO vehicles
- 3) Meet IHPRPT Phase II and III goals



**Monopropellant Thrust Stand**

(Economics of Space Transportation study (NASA-HQ) shows high energy density monoprops shrink vehicle size and cost for “generation-after-next” SSTO vehicles.)

# High Performance, Reduced Toxicity Monopropellants

Properties	RK-618A	RK-315E	RK-100A	Hydrazine
Density lsp, % over SOTA (a)	+58	+61	+24	0.0
Chamber Temp. (Theoretical) , K	2070	2083	1369	883
Carbon Content of Exhaust; (b)	none	none	none	none
Impact Sensitivity*, kg-cm (5 negatives)	>200	60	>200	>200
Friction Sensitivity, N (5 negatives)	318	300	>371	>371
NOL Card Gap (at 69 Cards)	negative	negative	negative	negative
Thermal Stability, % wt loss/48hr, 75°C	< 0.5	1.96	10.2	(< 0.1)
Melt Point, °C	5 (c)	<-22	-39	1

a: Theoretical, calculated with 300 psi chamber pressure, exhaust to vacuum, 50/1 expansion

b: as soot or solid carbon (by theoretical computation)

c: by DSC; melt transition was broad, melt peak reported

\*: For reference, n-propylnitrate had an impact sensitivity of 8 kg-cm

**RK-618 and RK-315 Propellants Display Acceptable  
Safety/Sensitivity Properties For Continued Development**

# **Advanced Monopropellants**

Dr. Tom Hawkins, Dr. Greg Drake, Adam Brand,

Milton McKay, Dr. Ismail Ismail

## **Recent Accomplishments**

## **Significance**

- |  |   |
|--|---|
| • Thruster test of low-toxicity monopropellant at National Hover Test Facility at PR-w and at Atlantic Research Corporation. Determined catalyst behavior, and C*, exhaust products. | • Technology transfer and evaluation in commercial industry of monopropellant with theoretical density impulse increase of 61% over hydrazine. Verifies in-house thruster test results. |
| • Synthesized and tested a dozen new ingredients. Identified a compound for scale-up.  | • Theoretical monopropellant performance with new ingredients equals or exceeds current bipropellant (NTO/MMH) systems.   |
| • Toxicology assessment, completed by AFRL/HE (Dr. Dave Mattie), shows these monoprops 6X less oral toxicity than hydrazine. Very low dermal irritation.                             | • Verifies that these propellants will have cost savings in handling, production, logistics, and disposal. Environmentally benign as well.  |
| • Gelled monopropellant developed.   | • Increases range of operability for some systems.  |
| • Numerous collaborators--Primex, ARC, NASA/GRC, small businesses...   | • Sharing resources will advance the program at a faster rate.  |

# **Liquid Hydrocarbon Fuels**

## **Objective:**

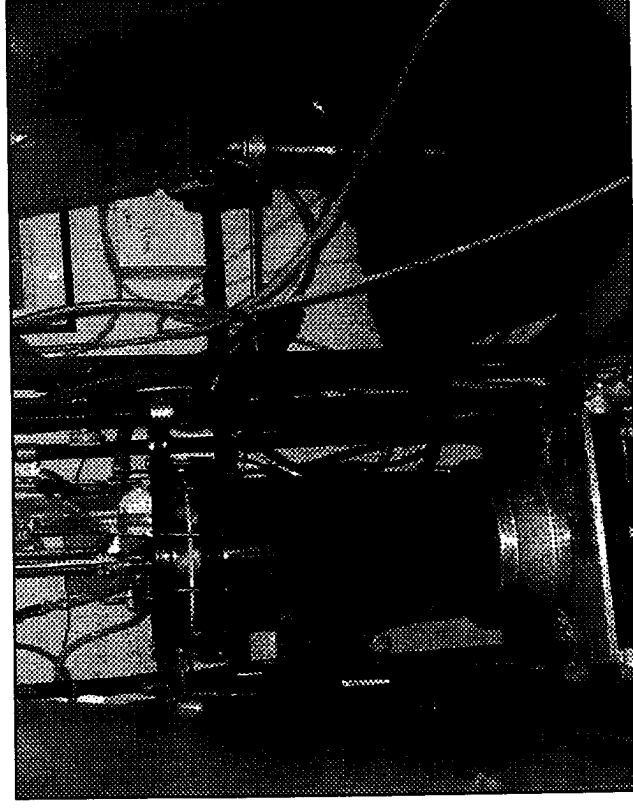
Develop cost-effective high energy, high density liquid hydrocarbons to enhance or replace RP and JP fuels.

## **Approach:**

- 1) HEDM synthesis; novel molecules; property testing
- 2) 10 lb. scale-up of promising candidates; combustion testing
- 3) Scale up to ton quantities through business collaborations

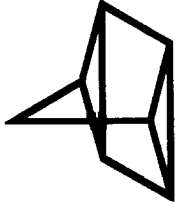

## **Motives:**

- 1) Enable new AF missions (air breathers or rockets)
- 2) NASA Access to Space concepts--supported with \$200K in FY99; \$50K likely in FY00
- 3) Meet IHPRT Phase II and III goals



**30L scale-up equipment at AFRL**

# Theoretical Performance Comparison of Energetic Hydrocarbons

Hydrocarbons	H/C ratio	Density g/ml	Calc.ΔHf (Kcal/mole)	Calc.Isp sec	
RP-1	1.9	0.80	-	300.0	
Quad	1.14	0.98	72.2	307.0	
 BCP	1.33	0.85	76.1	312.5	
AFRL-1	1.2	0.77	64.0	311.3	
AFRL-2	1.25	0.87	73.4	307.2	
AFRL-3	1.0	0.93	123.6	307.2	
AFRL-4	1.0	-	129.6	321.4	

# Energetic Hydrocarbon Fuels

Dr. Suresh Suri, Mike Tinnirello, Paul Jones

## Recent Accomplishments

- Four energetic hydrocarbons (BCP, AFRL-1, quadricyclane, 1,7 octadiyne) passed physical property, material compatibility, and thermal stability tests.
- Improved bicyclopopylidene (BCP) bench-scale synthesis to produce ~8 lbs.
- Delivered three energetic hydrocarbons (BCP, quadricyclane, 1,7 octadiyne) to NASA/MSFC for combustion testing.
- Initial tests by AFRL/HE determined quad and BCP to have low toxicity.
- BCP shown to be hypergolic with N<sub>2</sub>O<sub>4</sub>, IRFNA and possibly H<sub>2</sub>O<sub>2</sub>.
- Synthesized AFRL-3.

## Significance

- Candidates have passed initial hurdles. All are suitable for combustion testing.
- First time BCP produced in these quantities. Material now available for testing.
- Combustion results will indicate candidates potential for further scale-up.
- Verifies that these propellants can be handled in the same way as RP-1.
- BCP could be lower toxicity replacement for MMH. May enable new design of systems that burn nonhypergolic propellants.



# Summary

- Cryogenic solid HEDM doped with energetic atoms and molecules is being pursued with special attention on how to produce and measure 5% dopants in solid  $H_2$ .
- Synthesis of the  $N_5$  cation was a breakthrough towards making a high energy all nitrogen propellant.
- Low toxicity high performance monopropellants offer dramatic improvements over hydrazine.
- Fruitful collaboration with NASA/MSFC; will test fire BCP, 1,7-octadiyne, quadricyclane in 100-lb. thruster; Hydrocarbon program has products close to transition.